# REPORT DOCUMENTATION PAGE

dathering and mai	intaining the data need	ded and completing a	on is estimated to average 1 hond reviewing the collection of in	formatan Cand and			
1215 Jefferson D	ding suggestions for relativistic Highway, Suite	1204, Arlington, VA	to Department of Defense, Was 22202-4302. Respondents shi	hington Headquarter	rs Services, D notwithstandi	Directorate for Information Operations and Reports (0704-0188 ing any other provision of law, no person shall be subject to an	
PLEASE DO N	NOT RETURN YO	OUR FORM TO T	HE ABOVE ADDRESS.	valid OMB control nu	umber.		
1. REPORT DATE (DD-MM-YYYY) 2. REPORT TYPE				***************************************		3. DATES COVERED (From - To)	
	7-07-2007		Final			Apr03-Dec-06	
4. TITLE AND					5a. CO	NTRACT NUMBER	
The Subpipe Project							
					5b. GR	ANT NUMBER	
						F49620-03-1-0240	
					5c. PROGRAM ELEMENT NUMBER		
6. AUTHOR(S)					E4 DDC	DJECT NUMBER	
Bradshaw, P					Sa. Pho	DJEC1 NUMBER	
Diadsilan, 1	cici						
					5e. TASK NUMBER		
					5f. WORK UNIT NUMBER		
7 PERFORMI	NG ORGANIZAT	TION NAME(S) A	ND ADDRESS(ES)			8. PERFORMING ORGANIZATION	
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Peter Bradshaw						REPORT NUMBER	
	aw Engineering De	ent.				TAAAC, 26848	
Stanford, CA 94305-3032							
			ME(S) AND ADDRESS(ES			10. SPONSOR/MONITOR'S ACRONYM(S)	
~~~	0	<	ETENI Pa	malalph	584	-	
	AFOSRINA AVUNGTON UA  11. SPONSOR/MONITOR'S REPORT NUMBER(S)						
AFD:SRIAIA AVUNGTON OF 11. SPONSOR/MONITOR'S REPORT NUMBER(S)							
J	7100	AND		2220	5	NOMBEN(3)	
				***************************************			
12. DISTRIBUTION/AVAILABILITY STATEMENT Appriored Control for advance:							
distribution unlimited							
the section of the second section of the s							
13. SUPPLEMENTARY NOTES							
14. ABSTRAC	r						
See attached	report						
15. SUBJECT	TERMS						
16. SECURITY CLASSIFICATION OF: 17. LIMITATION OF 18. NUMBER 19a. NAME OF RESPONSIBLE PERSON						E OF RESPONSIBLE PERSON	
a. REPORT   b. ABSTRACT   c. THIS PAGE   ABSTRACT   OF				PAGES	Peter Bradshaw		
U	U	U	UU		19b. TELE	PHONE NUMBER (Include area code)	
				2		650-723-2968	

# AFOSR grant F49620-03-1-0240 Stanford SPO TAAAC, 26848 The Subpipe Project

Final Report, July 2007

Peter Bradshaw (Emeritus Professor) P.I. M.E. Dept., Stanford University Stanford, CA 94305-3032 bradshaw@stanford.edu

# Background

The purpose of this project was to examine the effect of surface roughness on a turbulent wall flow, in the limit of small roughness. It is, in effect, an attempt to throw light on experimental results dating from 1933, which appear to have been misinterpreted ever since as the result of an imprecise statement in the original (German) report by Nikuradse (a.k.a. Nikuradze) and its translations<sup>1</sup>. It was stated that below a certain Reynolds number  $u_\tau k/\nu$ , based on the (sand-grain) roughness height k and the friction velocity  $u_\tau$ , the roughness had "no effect" on the friction factor (i.e. no effect on the additive constant in the logarithmic law for velocity). All that can be deduced from the plotted results is that the effect became smaller than the experimental scatter, and there is no reason to suppose that the roughness effect depends on instability above some "critical" Reynolds number - particularly as later estimates of this Reynolds number are very low  $(u_\tau k/\nu)$  in the range of about 2 to 5).

The present author<sup>2</sup> produced an unrigorous but plausible argument that the roughness effect should vary as  $(u_{\tau}k/\nu)^2$  in the limit of small roughness. Briefly, the Reynolds number that matters is likely to be that based on the mean velocity at the height of the roughness (nominally  $k\tau_w/\mu$  if k lies within the linear part of the viscous sublayer) and on k itself, in which case it will vary as  $k^2$ .

In order to manufacture surfaces with small roughness Reynolds number repeatably, a closed-loop pipe rig was constructed, to be run at static pressures below atmospheric (hence the name "Subpipe") to obtain low unit Reynolds numbers, so that physically-large roughness elements could be used. The practical application of the work is to the development of a more precise criterion for allowable roughness on (pressure) wind-tunnel models. Final hand polishing of a model is time-consuming and expensive, so unnecessarily fine polishing is to be avoided.

#### Present Position

The project has been dogged by "bad luck". Shortly after the test rig (20 ft long) was

1

20080527159

completed, the P.I. had a bicycle accident, resulting in a broken hip, severe muscle damage and restricted mobility which has persisted to some extent. This delayed the start of experimental work for over a year. Then the rig was out of action for some time because it was inaccessible behind a wind tunnel being used for an unrelated experiment which lasted much longer than expected. Finally an alternative site was found, but it was found that the rig and its motor power supply had been damaged. Repairs took a long time, and the result is that no coherent results have yet been obtained.

The project's Stanford account has been closed. I have every intention of finishing this work: my general health remains good. Further expenditure will be borne by the P.I.'s personal Discretionary Fund.

## Note

In my letter of 5 Dec. 2006 (attached) I requested a further extension of the grant, in view of the unavoidable delays to the project. Our contracts office passed on an oral message that this request was denied but I have never received written notification, so there may be a gap in somebody's files.

## References

- 1. J. Nikuradse, Laws of flow in rough pipes. VDI Forschungsheft no. 361 (1933): trans.NACA TM 1292 (1950).
- 2. P. Bradshaw, A note on "critical roughness height" and "transitional roughness". Phys. Fluids  ${f 12},\ 1611$  (2000).